



1 **KEYWORDS:**

2 A: Discontinuous reinforcement; C: Analytical modelling; D: Mechanical testing; E:  
3 Compression moulding

4 **1. INTRODUCTION**

5 The mechanical properties of composites are generally governed by the mechanical  
6 properties of their constituents, fibre orientation and the fibre/matrix interface. Fibre  
7 orientation, which can be subdivided into three categories; aligned, partially aligned and  
8 randomly aligned, is generally represented using a fibre orientation factor ( $K_o$ ), according  
9 to the degree of fibre alignment relative to the loading direction. Composites with fibres  
10 aligned parallel to the loading direction are found to have the greatest composite strength  
11 and stiffness **as a result of effective stress transfer between fibre and matrix. Higher  $K_o$  also**  
12 **allows higher fibre volume fraction to be obtained** [1]. Natural fibre composites with a  
13 high degree of fibre alignment can be produced using carding, where long fibres are carded  
14 and combined with polymer prior to compression moulding. A high degree of fibre  
15 alignment can also be obtained using continuous fibre yarn, which is normally produced by  
16 fibre spinning or wrapping fibres together. Fibre yarns can be made from the same type of  
17 fibre, a mixture of different fibres, or can be a mixture of fibres and thermoplastic  
18 filaments [2]. Production of these forms of fibres, however, are often time consuming and  
19 limited to certain types of fibres. Some degree of fibre alignment can be obtained during  
20 injection moulding, where discontinuous fibres are generally added to the matrix in a loose  
21 form during extrusion prior to injection moulding, dependent on matrix viscosity, mould  
22 design and size, length and fibre concentration [3-5]. The least fibre orientation occurs in  
23 randomly aligned fibre mat made using hot pressing with polymeric powder or sheet.  
24 Composites with partially and random fibre alignment made using injection and  
25 compression moulding have greater ease of fabrication and often require less time to

1 produce, but their strength and stiffness are much lower due to poor fibre alignment. The  
2 interfacial shear strength ( $\tau$ ) between fibres and matrix determines the extent of stress  
3 transferred to fibres. A strong interface provides composites that display good strength and  
4 stiffness but tend to be brittle, whereas a weaker interface reduces the stress transfer from  
5 the matrix to the fibre, hence the composite displays lower strength and stiffness, but could  
6 be higher in composite toughness. It is acknowledged that due to its hydrophilic nature,  
7 cellulose based natural fibre [6-8] has limited interaction with hydrophobic matrices  
8 commonly resulting in limited interfacial strength.

9

10  $K_o$  and  $\tau$  are difficult to measure experimentally for fibres with irregularly shaped fibre  
11 cross-sections, particularly with fibre aggregation (found in most natural fibre composites).  
12 Alternatively, the Bowyer-Bader model, based on a modification of the Kelly-Tyson model  
13 can be used [9, 10]. The Bowyer-Bader model defines composite strength in terms of  
14 strength contribution from fibres of different length and the matrix, but if composite  
15 strength and fibre length distribution are known,  $K_o$  as well as  $\tau$  can be calculated. This  
16 model has been used to determine  $K_o$  and  $\tau$  for composites reinforced with discontinuous  
17 fibre prepared mostly using injection moulding and found to give good approximations  
18 relative to the experimental values.

19

20 The main objective of this paper was to evaluate the fibre orientation factors ( $K_o$ ) using the  
21 Bowyer-Bader model and compare with  $K_o$  values obtained for other discontinuous fibre  
22 composites.  $K_o$  values obtained here can also be used to better predict the composite  
23 strength and stiffness using theoretical modelling equations compared to commonly  
24 assumed values of  $K_o$  [11-16]. In this work, composites were reinforced with aligned  
25 discontinuous alkali treated harakeke and hemp fibre mats giving fibre contents for both

1 harakeke and hemp from 5 to 40 wt%, fibre orientation within mats was improved using a  
2 dynamic sheet former (DSF).  $\tau$  values obtained using the same model are also reported.

### 3 **2. THEORETICAL MODELLING - BOWYER - BADER MODEL**

4 As previously mentioned, the Bowyer and Bader predicts the tensile strength of  
5 discontinuous fibre composites of varying fibre alignment by considering the sum of the  
6 contributions of different fibre lengths, as well as that of the matrix [9, 10]. The Bowyer-  
7 Bader model can be expressed as follows:

$$\sigma_c = K_o(X + Y) + Z \quad (1)$$

8 where  $\sigma_c$  is the composite strength,  $Z$  the is strength contribution of the matrix,  $X$  and  $Y$  are  
9 the strength contributions of fibres with subcritical and supercritical fibre lengths  
10 respectively, relative to a critical fibre length determined using the following equation:

$$L_c = \frac{\sigma_f D}{2\tau} \quad (2)$$

11 such that,  $L_c$  is the fibre length required for the maximum stress in the fibre to reach the  
12 fibre fracture stress ( $\sigma_f$ ), i.e. fibre strength, and  $D$  is the average diameter of the fibre. As  
13 previously mentioned, this model can also be used to estimate fibre orientation factor ( $K_o$ )  
14 and interfacial shear strength ( $\tau$ ) if the composite strength and fibre length distribution are  
15 known. This model is found to give a good approximation relative to the values determined  
16 experimentally [10, 12, 17-20]. For the calculation of  $K_o$  and  $\tau$ , subcritical and supercritical  
17 fibre lengths are defined as the length of fibre shorter or longer relative to the critical fibre  
18 length at any particular strain ( $L_\epsilon$ ), as given by [9]:

$$L_\epsilon = \frac{E_f \epsilon_c D}{2\tau} \quad (3)$$

19 such that,  $E_f$  is the Young's modulus of the fibre,  $L_\epsilon$  is the fibre length, where, at particular  
20 strain, the fibre stress reaches a maximum only at the centre of the fibre. The average stress

1 in the fibres of length equal to or shorter than  $L_\epsilon$  at a particular strain according to the  
 2 Bowyer-Bader model is given by:

$$\sigma_{f\epsilon} = \frac{L\tau}{D} \quad (4)$$

3 and for the fibres longer than  $L_\epsilon$  is given by:

$$\sigma_{f\epsilon} = E_f \epsilon_c \left( 1 - \frac{E_f \epsilon_c D}{4L\tau} \right) \quad (5)$$

4 giving the overall composite stress at any strain level as:

$$\sigma_\epsilon = K_\theta \left[ \sum_i^{L_i < L_\epsilon} \frac{\tau L_i V_i}{D} + \sum_j^{L_j > L_\epsilon} E_f \epsilon_c V_j \left( 1 - \frac{E_f \epsilon_c D}{4\tau L_j} \right) \right] + E_m \epsilon_c (1 - V_f) \quad (6)$$

5 where  $E_m$  is the Young's modulus of the matrix,  $\sigma_\epsilon$  is the composite stress at a particular  
 6 strain,  $V_i$  and  $V_j$  are the volume fractions of the subcritical and supercritical fibre lengths,  
 7 respectively, and  $L_i$  and  $L_j$  are the subcritical and supercritical fibre lengths, respectively, in  
 8 reference to  $L_\epsilon$ . In order to determine  $K_\theta$  and  $\tau$ , the following steps were employed. Firstly,  
 9 values of two strains,  $\epsilon_{c1}$  and  $\epsilon_{c2}$  were selected from tensile stress-strain curve and the  
 10 corresponding composite stresses  $\sigma_{c1}$  and  $\sigma_{c2}$  were determined. Strength contributions of  
 11 the matrix ( $Z$ ) from a tensile test of neat PLA were also determined at the same strain  
 12 levels. These parameters ( $\sigma_{c1}$ ,  $\sigma_{c2}$  and  $Z$ ) were then used to calculate  $R$ , the ratio of the fibre  
 13 load bearing contribution using the following equation:

$$R = \frac{\sigma_{c1} - Z_1}{\sigma_{c2} - Z_2} \quad (7)$$

14 Then, an assumed value of  $\tau$  was taken and the corresponding values of  $L_{\epsilon1}$  and  $L_{\epsilon2}$   
 15 calculated using Equation (3). Accordingly, the fibre contribution terms  $X_1$ ,  $Y_1$  and  $X_2$ ,  $Y_2$   
 16 were calculated from  $X$  and  $Y$  (Equation (6)) for the different strain levels respectively, and  
 17 used to calculate  $R'$ , the theoretical value of  $R$  using the following equation:

$$R' = \frac{X_1 + Y_1}{X_2 + Y_2} \quad (8)$$

1 The assumed value of  $\tau$  was adjusted until  $R'=R$  and finally used in Equation (6) to obtain a  
2 value of  $K_\theta$  using strength and failure strain of the composite [21].

### 3 **3. EXPERIMENTAL METHODS**

#### 4 **3.1 Materials and Specimens**

5 Harakeke fibre treated with 5 wt% sodium hydroxide (NaOH) and 2 wt% sodium sulphite  
6 ( $\text{Na}_2\text{SO}_3$ ) and hemp fibre treated with 5 wt% sodium hydroxide (NaOH) were used to  
7 produce fibre mats [22]. Nature Works® 3052D injection moulding grade polylactide acid  
8 (PLA) with a density of 1250  $\text{kg/m}^3$  from Nature Works LLC, USA was used as a  
9 thermoplastic matrix.

10

11 Aligned fibre mats were produced using a Canpa automatic dynamic sheet former (DSF) as  
12 shown in Figure 1. In this study, fibre mats weighing 130 to 140  $\text{g/m}^2$  were produced using  
13 approximately 45 g treated discontinuous harakeke or hemp fibre. Firstly, fibre was  
14 dispersed in water (approximately 10 litres of water was used to disperse 10 g of fibre) in a  
15 mixing drum fitted with a disintegrator. Mats were then produced by spraying a controlled  
16 quantity of this low concentration suspension inside a spinning drum containing a porous  
17 medium to retain the fibre using a controlled pump and drum spinning speed.

18

19

**Figure 1**

20

21 The fibre mats were finally rinsed and compacted by allowing the drum to spin for a  
22 further 2 to 3 minutes. They were then dried in an oven at 80°C for 24 hours before being

1 cut to size (150mm x 90mm) to enable them to fit in a compression mould. Samples of  
2 dried harakeke and hemp fibre mats are shown in Figure 2.

3

4

## Figure 2

5

6 Fibre mats and PLA sheets were dried for at least 4 hours at 105°C and 60°C respectively.

7 They were weighed and arranged in a stack (in between Teflon sheets to prevent sticking

8 to mould) with relative numbers of each based on the required fibre weight percentage.

9 Stacks were heated and pressed in a hot press as for PLA samples (at 170°C and pressed

10 for 3 minutes at 3 MPa). After hot pressing, the moulded composite materials were

11 removed from the press and allowed to cool down to room temperature.

12

## 13 3.2 Methods

### 14 3.2.1 Scanning Electron Microscopy (SEM)

15 SEM was used to observe harakeke and hemp fibres. The observations were conducted

16 using a Hitachi S-4100 field emission scanning electron microscope operated at 3 kV.

17 Fibres were placed on aluminium stubs using double-sided adhesive tape and sputter

18 coated with platinum and palladium to make them conductive.

19

### 20 3.2.2 Light Microscopy

21 Transverse and longitudinal sections of composite samples were obtained for examination.

22 Moulds were initially wiped with a release agent to facilitate specimen removal and epoxy

23 resin was poured into the mould followed by placement of composite sections. Specimens

24 were demoulded after 24 hours of curing at room temperature and ground and polished

25 using a series of coarse and fine abrasive papers starting from 320 followed by 500, 1000,

1 2000 and 4000 grit for approximately three to five minutes for each step. Optical images of  
2 fibre mats and composite sections were taken using an Olympus BX60F5 optical light  
3 microscope.

### 4 3.2.3 Tensile Testing of PLA and Composites

5 Prior to tensile testing, all specimens were placed in a conditioning chamber at  $23^{\circ}\text{C} \pm 3^{\circ}\text{C}$   
6 and  $50\% \pm 5\%$  relative humidity for at least 48 hours. 5 replicate samples were tested for  
7 each batch and average tensile strength (TS) and Young's modulus (YM) were obtained  
8 using the results from all specimens. Tensile testing followed the procedures detailed in  
9 ASTM D 638-03; Standard Test Method for Tensile Properties of Plastics. An Instron  
10 2630-112 extensometer was attached to the central part of the test specimen to measure the  
11 specimen extension. The specimens were tested at a constant rate of 2 mm/min. Stress  
12 versus strain graphs were constructed for obtaining orientation factor using the Bowyer-  
13 Bader model. Further details of mechanical performance are published separately.

## 14 4. RESULTS AND DISCUSSION

### 15 4.1 Assessment of Fibre Morphology

16 Figure 3 shows the surface of alkali treated harakeke and hemp fibres. Hemp fibre  
17 appeared to have smoother surface, a less rounded shape and have larger fibre diameter  
18 compared to harakeke fibre. From Figure 4, it is evident that the average diameter for  
19 hemp fibre is significantly larger with greater standard deviation relative to harakeke fibre.  
20 Although, the average tensile for hemp fibre is higher [22], the contribution of fibre  
21 strength towards composite tensile strength for hemp fibre is expected to be smaller as a  
22 result of smaller surface area.  
23  
24  
25



1 **Figure 3**

2  
3  
4 **Figure 4**

5  
6  
7  
8 **4.2 Assessment of Fibre Orientation**

9 Representative images of cross-sections of harakeke and hemp composites with 20 wt%  
10 fibre content sectioned parallel and perpendicular to the DSF rotation direction are shown  
11 respectively in Figures 5 and 6 (parallel (a), perpendicular (b)). Fibres sectioned  
12 longitudinally appear as dark lines whereas those sectioned transversely appear as light  
13 circles. It is clearly evident that the fibres have reasonable alignment in the DSF rotation  
14 direction as more transverse fibre cross-sections were observed within the sections cut  
15 perpendicular to DSF direction and more longitudinal fibre sections were observed within  
16 the sections cut parallel to the DSF rotation direction. It can also be observed that the fibres  
17 are well dispersed without obvious fibre agglomeration. Similar orientation was seen at 30  
18 wt% content (Figures 7 and 8); however, at this fibre content, there are some fibre  
19 agglomerates and voids in the hemp composites. Comparison of fibre agglomerations for  
20 harakeke and hemp composites reinforced with 20 and 30 wt% are shown in Figure 9. It  
21 can be seen that, the fibre agglomeration for composites with 30 wt% fibre is more  
22 noticeable compared to that with 20 wt% fibre. Fibre agglomeration is attributed to  
23 insufficient polymer for adequate wetting increased possibility of fibre-fibre interaction.  
24 The formation of voids at high fibre content could be due to the evaporation of a greater  
25 amount of moisture from the fibre which would be expected to reduce the effect of  
26 reinforcement.

27  
28 **Figure 5**

1  
2  
3  
4  
5  
6  
7  
8  
9

**Figure 6**  
  
**Figure 7**  
  
**Figure 8**  
  
**Figure 9**

10 **4.3 Determination of Fibre Orientation Factor ( $K_\theta$ ) and Interfacial Shear Strength ( $\tau$ )**  
11 **According to the Bowyer - Bader Model**

12 In this work, as previously mentioned, the Bowyer-Bader model has been employed to  
13 calculate fibre orientation factor ( $K_\theta$ ) and interfacial shear strength ( $\tau$ ).  $K_\theta$  and  $\tau$  were  
14 determined using the experimental tensile stress-strain curves and the fibre length  
15 distributions, following the procedure detailed previously. Most harakeke and hemp fibre  
16 composites failed with failure strains of less than 0.02 and so composite stresses ( $\sigma_1$  and  
17  $\sigma_2$ ) corresponding to strain values well away from final fracture strain,  $\varepsilon_1 = 0.005$  and  $\varepsilon_2 =$   
18  $0.01$ , such that  $\varepsilon_2 = 2\varepsilon_1$  were selected (as required by the model) [23]. Analysis was also  
19 conducted using strains of  $\varepsilon_1 = 0.0075$  and  $\varepsilon_2 = 0.015$  for comparison. Using higher strain  
20 levels has been shown to give lower values of  $K_\theta$ , as reinforcement efficiency decreased  
21 due to matrix cracking and fibre debonding occurring closer to the composite failure strain  
22 [10]. Typical composite stress-strain curves for harakeke and hemp fibre composites  
23 obtained from tensile testing are shown in Figure 10. The curves correspond to the  
24 experimental results that were close to the composites mean tensile strength values. It is  
25 well accepted that stress-strain curves of many thermoplastic composites are non-linear

1 even at low strain levels. Due to this non-linearity, polynomial curve fitting parameters  
2 have been used to improve the accuracy of the analysis [12].

3  
4 **Figure 10**

5  
6 The relationship of composite stresses at strains of 0.005 and 0.01 versus fibre content is  
7 shown in Figure 11; stress can be seen to increase with increasing fibre content as expected  
8 from Equation (6). A linear regression line is used to fit the data and shows good  
9 agreement with the data. However, it may be observed that at 30 wt% fibre content, the  
10 reinforcement efficiency is slightly decreased.

11  
12 **Figure 11**

13  
14 The analysis was conducted by substituting the following parameters into the model:  $E_f =$   
15 21.2 GPa for harakeke and 26.4 GPa for hemp,  $E_m = 3.6$  GPa for PLA,  $\sigma_f = 782.4$  MPa and  
16 911.3 MPa for harakeke and hemp respectively, as obtained in previous work [22]. It was  
17 found that at low strain levels,  $L_\epsilon$  is small relative to the (measured) fibre lengths and so  
18 most of the length of fibres would be at the value of maximum stress. Therefore, the  
19 composite is tending towards behaving as a composite reinforced with continuous fibres  
20 [19].  $K_\theta$  and  $\tau$  and other related parameters obtained for composites at different fibre  
21 contents are tabulated in Table 1.

22 **Table 1**

23  
24 It can be seen from the table that  $K_\theta$  for harakeke fibres were higher than for hemp fibres at  
25 all fibre contents.  $K_\theta$  for both harakeke and hemp fibre composites were higher than values

1 seen in the literature for those composites prepared using injection moulding and hot  
2 pressed using randomly oriented fibre mats (0.24 – 0.38) [17, 18, 20, 24, 25] and slightly  
3 lower compared to the highest values obtained with aligned flax/polypropylene nonwoven  
4 preforms (0.45 – 0.6) composites [26]. The highest  $K_o$  values for harakeke and hemp in this  
5 work were found to be 0.58 (15 wt %) and 0.44 (20 wt %) respectively. Lower  $K_o$  values  
6 for hemp fibre composites compared to harakeke fibre composites could be related to  
7 lower aspect ratio and fibre agglomeration influencing orientation. It should be noted that  
8 the contribution of fibres and matrix (X, Y and Z) are based on intrinsic properties while the  
9 experimental composite strength is based on numbers of mechanisms, i.e. fibre length,  
10 fibre orientation, interfacial bonding and fibre debonding. Fibres with length longer than  
11 critical fibre length as for harakeke contribute more towards the actual strength of  
12 composite, thus higher  $K_o$  can be obtained. Similarly, if degree of fibre agglomeration is  
13 lower, thus the resulting experimental composite strength is higher, hence higher  $K_o$ . A  
14 similar trend has been observed in previous work, where decreased fibre length for  
15 composites at high fibre fractions can cause a decreased  $K_o$  in injection moulded samples  
16 [17, 25, 27]. Higher  $K_o$  values for injected moulded glass and carbon fibre composites than  
17 for injection moulded natural fibre composites have been obtained, which fits with the  
18 expectation that a higher degree of fibre straightness along with less possibility of fibre  
19 agglomeration can also contribute to an increase in  $K_o$  [9, 10, 12, 19, 28].

20

21 From our data, it was also found that the  $K_o$  obtained was not entirely independent of  
22 strain, in contrast with the assumption used in the model [9, 10];  $K_o$  generally decreased at  
23 higher strain levels closer to composite failure, believed to be due to matrix cracking and  
24 fibre debonding as previously mentioned, particularly for hemp fibre composites for which  
25 the higher strain used was closer to the composite failure strain (indeed some had failed at

1 strains lower than those used in analysis as reflected by a "-" in table) [18]. From Figure  
2 12, it is also evident that  $K_o$  decreased as fibre content increased. A similar trend has also  
3 been observed in previous work for injection moulded composites [17, 25, 27]. Reduction  
4 in  $K_o$  for harakeke and hemp composites with fibre contents of more than 30 wt% fibre is  
5 more evident, corresponding to a higher potential of fibre misalignment during processing  
6 due to fibre agglomeration as seen in Figures 7 and 8, which would be more likely at  
7 higher fibre contents due to fibre–fibre interaction from hydrogen bonding mentioned  
8 previously [29]. It should be noted that fibre alignment would also be expected to depend  
9 on matrix viscosity during processing, such that polymers with higher viscosity would  
10 cause a higher degree of fibre displacement as a result of higher pressure used during  
11 processing.

12  
13 **Figure 12**  
14

15 General trends for  $\tau$  versus fibre content for harakeke and hemp fibre composites are  
16 graphically presented in Figure 13.  $\tau$  values obtained were found to be comparable to  
17 values reported in previous studies [30-36]. Linear-trend lines fitted to the predicted  $\tau$   
18 values suggest a slight increase with increased fibre content. Increase in  $\tau$  values could be  
19 explained by the increase in pressure used at higher fibre contents increasing interfacial  
20 bonding [19, 25]. **Notably,  $\tau$  values for harakeke and hemp composites with 20 wt% fibre**  
21 **were found to be the highest, thought to be due to the greatest fibre contribution towards**  
22 **composite strength as at this fibre content and degree of fibre distortion and agglomeration**  
23 **could be minimal.**

1 **Figure 13**

2  
3 It can be seen that  $\tau$  for harakeke was slightly higher than for hemp fibre, except at 20 wt%  
4 fibre, which could be related to the greater surface area for harakeke fibre due to smaller  
5 fibre diameter (see Figure 4). A higher  $\tau$  for harakeke fibre compared to hemp fibre could  
6 also be due to a higher degree of surface roughness for harakeke as can be seen in Figure 3.

7 **5. CONCLUSIONS**

8 The alignment of fibres within composites reinforced using fibre mats produced using a  
9 DSF has been evaluated. It was found that using a DSF to make fibre mats lead to better  
10 fibre alignment compared to that for composites made using injection moulding and  
11 composites reinforced with randomly oriented fibre mats. The orientation factors for  
12 harakeke and hemp fibre composites were found to be up to 0.58 and 0.44 respectively.  
13 Fibre orientation and interfacial shear strength for harakeke and hemp fibre composites  
14 were found to be slightly dependent on fibre content, believed to be due to fibre  
15 agglomeration and increased pressure within composite materials during processing.

16 **ACKNOWLEDGEMENT**

17 This research received no specific grant from any funding agency in the public,  
18 commercial, or not-for-profit sectors. However, the authors would like to thank to the  
19 Composites Research Group, University of Waikato for their support and the Ministry of  
20 Higher Education and Universiti Teknologi Mara Malaysia for the scholarship.

## 1 REFERENCES

- 2 [1] Yu H, Potter K, Wisnom M. A novel manufacturing method for aligned discontinuous fibre  
3 composites (High Performance-Discontinuous Fibre method). *Composites Part A: Applied Science*  
4 *and Manufacturing*. 2014;65:175-85.
- 5 [2] Zhang L, Miao M. Commingled natural fibre/polypropylene wrap spun yarns for structured  
6 thermoplastic composites. *Composites Science and Technology*. 2010;70(1):130-5.
- 7 [3] Joseph P, Joseph K, Thomas S. Effect of processing variables on the mechanical properties of  
8 sisal-fiber-reinforced polypropylene composites. *Composites Science and Technology*.  
9 1999;59(11):1625-40.
- 10 [4] Chen C-S, Chen T-J, Chen S-C, Chien R-D. Optimization of the injection molding process for  
11 short-fiber-reinforced composites. *Mechanics of Composite Materials*. 2011;47(3):359-68.
- 12 [5] Beckermann G. *The Processing, Production and Improvement of Hemp-Fibre Reinforced*  
13 *Polypropylene Composite Materials: University of Waikato; 2004.*
- 14 [6] De Rosa I, Iannoni A, Kenny J, Puglia D, Santulli C, Sarasini F, et al. Poly (lactic acid)/Phormium  
15 tenax composites: Morphology and thermo-mechanical behavior. *Polymer Composites*.  
16 2011;32(9):1362-8.
- 17 [7] Le Guen MJ, Newman RH. Pulped Phormium tenax leaf fibres as reinforcement for epoxy  
18 composites. *Composites Part A: Applied Science and Manufacturing*. 2007;38(10):2109-15.
- 19 [8] Cruthers NM, Carr DJ, Laing RM, Niven BE. Structural differences among fibers from six  
20 cultivars of harakeke (Phormium tenax, New Zealand flax). *Textile Research Journal*.  
21 2006;76(8):601-6.
- 22 [9] Bader M, Bowyer W. An improved method of production for high strength fibre-reinforced  
23 thermoplastics. *Composites*. 1973;4(4):150-6.
- 24 [10] Bowyer W, Bader M. On the re-inforcement of thermoplastics by imperfectly aligned  
25 discontinuous fibres. *Journal of materials Science*. 1972;7(11):1315-21.
- 26 [11] Thomason J, Vlug M, Schipper G, Krikor H. Influence of fibre length and concentration on the  
27 properties of glass fibre-reinforced polypropylene: Part 3. Strength and strain at failure.  
28 *Composites Part A: Applied Science and Manufacturing*. 1996;27(11):1075-84.
- 29 [12] Thomason J. Micromechanical parameters from macromechanical measurements on glass  
30 reinforced polyamide 6, 6. *Composites Science and Technology*. 2001;61(14):2007-16.
- 31 [13] Sanadi A, Piggott M. Interfacial effects in carbon-epoxies. *Journal of materials Science*.  
32 1985;20(2):421-30.
- 33 [14] Thomason J, Vlug M. Influence of fibre length and concentration on the properties of glass  
34 fibre-reinforced polypropylene: 1. Tensile and flexural modulus. *Composites Part A: Applied*  
35 *Science and Manufacturing*. 1996;27(6):477-84.
- 36 [15] Fu S-Y, Lauke B, Mäder E, Yue C-Y, Hu X. Tensile properties of short-glass-fiber-and short-  
37 carbon-fiber-reinforced polypropylene composites. *Composites Part A: Applied Science and*  
38 *Manufacturing*. 2000;31(10):1117-25.
- 39 [16] Aziz SH, Ansell MP. The effect of alkalization and fibre alignment on the mechanical and  
40 thermal properties of kenaf and hemp bast fibre composites: Part 1–polyester resin matrix.  
41 *Composites Science and Technology*. 2004;64(9):1219-30.
- 42 [17] Serrano A, Espinach F, Julian F, Del Rey R, Mendez J, Mutje P. Estimation of the interfacial  
43 shears strength, orientation factor and mean equivalent intrinsic tensile strength in old  
44 newspaper fiber/polypropylene composites. *Composites Part B: Engineering*. 2013;50:232-8.
- 45 [18] Bos HL, Müssig J, van den Oever MJ. Mechanical properties of short-flax-fibre reinforced  
46 compounds. *Composites Part A: Applied Science and Manufacturing*. 2006;37(10):1591-604.
- 47 [19] Sui G, Wong S-C, Yang R, Yue C. The effect of fiber inclusions in toughened plastics–Part II:  
48 determination of micromechanical parameters. *Composites Science and Technology*.  
49 2005;65(2):221-9.

- 1 [20] López J, Boufi S, El Mansouri N, Mutjé P, Vilaseca F. PP composites based on mechanical pulp,  
2 deinked newspaper and jute strands: a comparative study. *Composites Part B: Engineering*.  
3 2012;43(8):3453-61.
- 4 [21] Lafranche E, Oliveira VM, Martins CI, Krawczak P. Prediction of injection-moulded flax fibre  
5 reinforced polypropylene tensile properties through a micro-morphology analysis. *Journal of*  
6 *Composite Materials*. 2013:0021998313514875.
- 7 [22] Efendy MA, Pickering K. Comparison of harakeke with hemp fibre as a potential  
8 reinforcement in composites. *Composites Part A: Applied Science and Manufacturing*.  
9 2014;67:259-67.
- 10 [23] Pickering K, Efendy MA. Preparation and mechanical properties of novel bio-composite made  
11 of dynamically sheet formed discontinuous harakeke and hemp fibre mat reinforced PLA  
12 composites for structural applications. *Industrial Crops and Products*. 2016;84:139-50.
- 13 [24] Beckermann G, Pickering KL. Engineering and evaluation of hemp fibre reinforced  
14 polypropylene composites: fibre treatment and matrix modification. *Composites Part A: Applied*  
15 *Science and Manufacturing*. 2008;39(6):979-88.
- 16 [25] Vallejos M, Espinach F, Julian F, Torres L, Vilaseca F, Mutje P. Micromechanics of hemp  
17 strands in polypropylene composites. *Composites Science and Technology*. 2012;72(10):1209-13.
- 18 [26] Miao M, Shan M. Highly aligned flax/polypropylene nonwoven preforms for thermoplastic  
19 composites. *Composites Science and Technology*. 2011;71(15):1713-8.
- 20 [27] Thomason J. The influence of fibre length and concentration on the properties of glass fibre  
21 reinforced polypropylene: 7. Interface strength and fibre strain in injection moulded long fibre PP  
22 at high fibre content. *Composites Part A: Applied Science and Manufacturing*. 2007;38(1):210-6.
- 23 [28] Thomason J. The influence of fibre length and concentration on the properties of glass fibre  
24 reinforced polypropylene. 6. The properties of injection moulded long fibre PP at high fibre  
25 content. *Composites Part A: Applied Science and Manufacturing*. 2005;36(7):995-1003.
- 26 [29] Pickering K, Abdalla A, Ji C, McDonald A, Franich R. The effect of silane coupling agents on  
27 radiata pine fibre for use in thermoplastic matrix composites. *Composites Part A: Applied Science*  
28 *and Manufacturing*. 2003;34(10):915-26.
- 29 [30] Bax B, Müssig J. Impact and tensile properties of PLA/Cordenka and PLA/flax composites.  
30 *Composites Science and Technology*. 2008;68(7):1601-7.
- 31 [31] Islam M, Pickering K, Foreman N. Influence of alkali treatment on the interfacial and physico-  
32 mechanical properties of industrial hemp fibre reinforced polylactic acid composites. *Composites*  
33 *Part A: Applied Science and Manufacturing*. 2010;41(5):596-603.
- 34 [32] Joffe R, Andersons J, Wallström L. Strength and adhesion characteristics of elementary flax  
35 fibres with different surface treatments. *Composites Part A: Applied Science and Manufacturing*.  
36 2003;34(7):603-12.
- 37 [33] Prajer M, Ansell MP. Interfacial micromechanics in PLA/sisal fibre composites. *Proceedings of*  
38 *the 11th International Conference on Non-Convertional Materials and Technologies*2009. p. 6-9.
- 39 [34] Sawpan MA, Pickering KL, Fernyhough A. Effect of fibre treatments on interfacial shear  
40 strength of hemp fibre reinforced polylactide and unsaturated polyester composites. *Composites*  
41 *Part A: Applied Science and Manufacturing*. 2011;42(9):1189-96.
- 42 [35] Torres F, Cubillas M. Study of the interfacial properties of natural fibre reinforced  
43 polyethylene. *Polymer Testing*. 2005;24(6):694-8.
- 44 [36] Zafeiropoulos NE. On the use of single fibre composites testing to characterise the interface  
45 in natural fibre composites. *Composite Interfaces*. 2007;14(7-9):807-20.
- 46